FTR - Flight Test Report

Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nicht auszugsweise, vervielfältigt werden.





Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	30.09.2016	Minimum take off weight Maximum take off weight 105 kg 130 kg				
Testpilot		According to the flighttests from the 30th	Sentember 2016, the following results			
Harness		According to the flighttests from the 30th September 2016, the following results have been confirmed				
Pilot's take off weight						

Classification C



Test-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation		
1. Inflation / take-off - 4.4.1							
Rising behavior		Easy rising, some pilot correction is required	В	Easy rising, some pilot correction is required	В		
Special take off technique required		No	Α	No	Α		
2. Landing - 4.4.2			'				
Special landing technique required		No	A	No	Α		
3. Speeds in straight flight - 4.4.3		-					
Trim speed more than 30km/h		Yes	A	Yes	Α		
Speed range using the controls larger than 10km/h	n	Yes	A	Yes	Α		
Minimum speed		25 km/h to 30 km/h	В	25 km/h to 30 km/h	В		
4. Control movement - 4.4.4							
Max. weight in flight up to 80kg			-		-		
Max. weight in flight 80 to 100kg	√lax. weight in flight 80 to 100kg		С	Increasing 45cm - 60cm	С		
Max. weight in flight greater than 100kg		-			-		
5. Pitch stability exiting accelerated flight - 4.4	1.5	•	'				
Dive forward angle on exit		Dive forward less than 30°	Α	Dive forward less than 30°	Α		
Collapse occurs		No	A	No	A		
6. Pitch stability operating controls during acc	elerated f	light - 4.4.6					
Collapse occurs		No	A	No	Α		
7. Roll stability and damping - 4.4.7							
Oscillations		Reducing	A	Reducing	A		
8. Stability in gentle spirals - 4.4.8		reducing	Α	reducing			
Tendency to return to straight flight		Spontaneous exit	A	Spontaneous exit	Α		
Behaviour exiting a fully developed spiral di	ivo 11	•	A	Орогнальной сих			
Initial response of glider (first 180°)	100 - 4.4	No immediate reaction	В	No immediate reaction	В		
Tendency to return to straight flight			A	Spontaneous exit	A		
Turn angle to recover normal flight			В	720° to 1080°, spontaneous recovery	В		
10. Symmetric front collapse - 4.4.10		720° to 1080°, spontaneous recovery	-				
Folding lines used		l No		l No			
Entry	*	Rocking back less than 45°	Α	Rocking back less than 45°	Α		
Recovery	%0€ ~ pe	Spontaneous in 3 to 5 sec	В	Spontaneous in 3 to 5 sec	В		
Dive forward angle on exit	98 dr	0° - 30° Entering a turn of less than	190° A	30° - 60° Keeping course	В		
Cascade occurs	trim	No	Α	No	Α		
Entry	% O9:	Rocking back less than 45°	Α	Rocking back less than 45°	Α		
Recovery	5 ^ P86 dr	Spontaneous in 3 to 5 sec	В	Spontaneous in 3 to 5 sec	В		
Dive forward angle on exit	Ę.	30° - 60° Entering a turn of less than		0° - 30° Keeping course	Α		
Cascade occurs		No	A	No	A		
Entry	86	Rocking back less than 45°	Α	Rocking back greater than 45°	С		
Recovery	ated v S	Spontaneous in 3 to 5 sec	В	Spontaneous in 3 to 5 sec	В		
Dive forward angle on exit	98	30° - 60° Entering a turn of 90° to 18		0° - 30° Entering a turn of less than 90°	Α		
Cascade occurs		No	Α	No	Α		
11. Exiting deep stall (parachutal stall) - 4.4.11		T v		I.v.			
Deep stall achieved		Yes		Yes			
Recovery		Spontaneous in less than 3 sec	A	Spontaneous in less than 3 sec			
Dive forward angle on exit		30° - 60°	В	30° - 60°			
Change of course		Changing course less than 45°	A	Changing course less than 45° No			
Cascade occurs		No	Α	No			

12. High angle of attack recovery - 4.4.12									
Recovery	Spontaneous in I	less than 3 sec		Α	Spontaneous in lose than 2 sec			Α	
Cascade occurs		less than 5 sec		A	Spontaneous in less than 3 sec				
13. Recovery from a developed full stall - 4.4.	No			А	No			Α	
Dive forward angle on exit	30° - 60°			В	30°-60°			В	
Collapse	No collapse			A	No collapse			A	
Cascade occurs (other than collapse) Rocking backward		No Less than 45°			A	No Less than 45°			A A
Line tension		Most lines tight			Α	Most lines tight			A
14. Asymmetric collapse (trim speed) - 4.4.14									
Folding lines used		No		Г		No	T		
Change of course until re-inflation	se	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-inflation			Α	Spontaneous re-	Spontaneous re-inflation		
Total change of course		Less than 360°			Α	Less than 360°			Α
Collapse on the opposite side occurs Twist occurs	trir nax 5	No No			A A	No No			A A
Cascade occurs	u	No No			A	No			A
Change of course until re-inflation	60	90° - 180°	Dive or roll angle	15° - 45°	В	< 90°	Dive or roll angle	45° - 60°	С
Do inflation behavior	trim speed, max 75% collapse	Inflatos in loss th	on 2 oon from et	art of pilot action	-	Coortonoous ro	inflation		Δ.
Re-inflation behavior	spee % col		nan 3 sec from sta	art of pilot action	С	Spontaneous re-inflation			A
Total change of course Collapse on the opposite side occurs	trim : x 75°	Less than 360° No			A	Less than 360° No			A
Twist occurs	EE .	No			Α	No			Α
Cascade occurs		No			A	No			Α
Change of course until re-inflation	9	90° - 180°	Dive or roll angle	15° - 45°	В	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	, pa	Spontaneous re-	inflation	1	Α	Spontaneous re-	inflation		Α
Total change of course	accelerated, max 50% collapse	Less than 360°	uuVII		A	Less than 360°	auuvii		A A
Collapse on the opposite side occurs	acce x 50	No			A	No			A
Twist occurs	e E	No			A	No			A
Cascade occurs		No non tone		450.000	A	No		450,000	A
Change of course until re-inflation	bse	90° - 180°	Dive or roll angle	45° - 60°	С	< 90°	Dive or roll angle	45° - 60°	С
Re-inflation behavior	accelerated, max 75% collapse	Inflates in less th	nan 3 sec from sta	art of pilot action	С	Spontaneous re-	-inflation		Α
Total change of course	celer	្គឹ Less than 360°		Α	Less than 360°			Α	
Collapse on the opposite side occurs Twist occurs	ao i xer	No			A	No No			A A
Cascade occurs	Ë No No			A	No No			A	
15. Directional control with a maintained asym	5. Directional control with a maintained asymmetric collapse - 4.4.15								
Able to keep course straight	Yes			Α	Yes			Α	
180° turn away from the collapsed side possible in	Yes			Α	Yes			Α	
Amount of control range between turn and stall or	25% to 50% of the symmetric control travel			С	25% to 50% of the symmetric control travel			С	
16. Trim speed spin tendency - 4.4.16	-								
Spin occurs		No			Α	No			Α
17. Low speed spin tendency - 4.4.17									
Spin occurs		No			Α	No			Α
18. Recovery from a developed spin - 4.4.18		ı				I			
Spin rotation angle after release		Stops spinning in less than 90°		Α	Stops spinning in 90° to 180°			С	
Cascade occurs		No			Α	No	Α		
19. B-line-stall - 4.4.19 Change of course before release		Changing course	loss than 45°		Α	Changing course	a lose than 45°		А
Behaviour before release		Changing course less than 45° Remains stable without straight span		C	Changing course less than 45°			C	
					Remains stable without straight span				
Recovery		Spontaneous in less than 3 sec		Α	Spontaneous in 3 to 5 sec			В	
Dive forward angle on exit Cascade occurs		30° - 60° No			A	0° - 30° No			A
20. Big ears - 4.4.20									
Entry procedure		Standard technic	oue		Α	Special device r	equired		Α
Behaviour during big ears	Stable flight			A	Stable flight			A	
Recovery		Spontaneous in 3 to 5 sec		В	Recovery through pilot action in less than a furth		ess than a further	В	
Dive forward angle on exit		O° - 30°		A	3 sec			A	
21. Big Ears in accelerated flight - 4.4.21	U - 3U				0° bis 30°				
Entry procedure	Standard technique			Α	Standard technique			Α	
Behaviour during big ears	Stable flight		A	Stable flight			A		
Recovery	Spontaneous in 3 to 5 sec		A	Recovery through pilot action in less than a further			В		
Dive forward angle on exit		0°-30°		Α	3 sec 0° bis 30°			Α	
Behaviour immediately after releasing the accelar maintaining big ears	Stable flight		Α	Stable flight			Α		
23. Alternative means of directional control - 4	1.4.22								
180° tum achievable in 20 sec	Yes		Α	Yes			Α		
Stall or spin occurs	No No			A	No No			A	
23. Any other flight procedure and/or configura	ation desc		's manual - 4.4.	23					
Procedure works as descibed					NA				NA
Procedure suitable for novice pilots Cascade occurs				NA NA				NA NA	
Cascade occurs									
24. Remarks of testpilot:									