FTR - Flight Test Report Dieser Prütbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nic

Manufacturer		Type testing No.	EAPR-GS-0765/18		
	UP International Kreuzeckbahnstraße 7 D-82462 Garmisch-Partenkirchen	serial number	XA-61L-03-1-108-7376		
Model	Ascent-4 L	Location	Bassano		
Comment		Location	Schlick 2000, Stubai		



Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	10.01.2018	Minimum take 95 kg	reight	Maximum take off weight 130 kg			
Testpilot		Pascal Purin			Anselm Rauh		
Harness	arness		Bassano		EAPR		
Pilot's take off weigh	nt	95	kg		128	kg	

Classification



est-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation	
1. Inflation / take-off - 4.4.1						
Rising behavior		no pilot correction required	А	no pilot correction required	А	
Special take off technique required		No	Α	No	Α	
2. Landing - 4.4.2		140		140		
		No		Ne		
Special landing technique required		INO	Α	No	Α	
3. Speeds in straight flight - 4.4.3						
Trim speed more than 30km/h		Yes	Α	Yes	A	
Speed range using the controls larger than 10km/	h	Yes	Α	Yes	Α	
Minimum speed		Less than 25 km/h	Α	Less than 25 km/h	Α	
4. Control movement - 4.4.4						
Max. weight in flight up to 80kg			-		-	
Max. weight in flight 80 to 100kg			-		-	
Max. weight in flight greater than 100kg		Increasing >65 cm	А	Increasing >65 cm	А	
5. Pitch stability exiting accelerated flight - 4.	4.5	•				
Dive forward angle on exit		Dive forward less than 30°	А	Dive forward less than 30°	А	
Collapse occurs		No	Α	No	Α	
6. Pitch stability operating controls during acc	celerated	flight - 4.4.6				
Collapse occurs		No	Α	No	Α	
7. Roll stability and damping - 4.4.7						
Oscillations		Reducing	Α	Reducing	А	
8. Stability in gentle spirals - 4.4.8						
Tendency to return to straight flight		Spontaneous exit	l A	Spontaneous exit	A	
9. Behaviour exiting a fully developed spiral d	live - 4.4.			- Committee of the comm		
Initial response of glider (first 180°)	itial response of glider (first 180°)		Α	Immediate reduction of rate in turn	Α	
endency to return to straight flight		Spontaneous exit	Α	Spontaneous exit	Α	
Furn angle to recover normal flight		Less than 720°, spontaneous recovery	Α	Less than 720°, spontaneous recovery	Α	
10. Symmetric front collapse - 4.4.10						
Folding lines used		No		No		
Entry	30%	Rocking back less than 45° A Rocking back less than 4		Rocking back less than 45°	Α	
Recovery	.0€ ~ peeds	Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	trim sp	0° - 30° Keeping course	Α	0° - 30° Keeping course	Α	
Cascade occurs		No	Α	No	Α	
Entry	%0g <	Rocking back less than 45°	Α	Rocking back less than 45°	A	
Recovery	, < paeds	Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	ili so	0° - 30° Keeping course	Α	0° - 30° Keeping course	A	
Cascade occurs	Σ	No	A	No	A	
Entry	% %	Rocking back less than 45°	Α	Rocking back less than 45°	A	
Recovery	ated > 50%	Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	xele	0° - 30° Keeping course	Α	0° - 30° Keeping course	Α	
Cascade occurs	ä	No	Α	No	A	
11. Exiting deep stall (parachutal stall) - 4.4.1	1					
Deep stall achieved		Yes		Yes		
Recovery	overy		s than 3 sec A Spontaneous in less than 3 sec		Α	
Dive forward angle on exit		0° - 30°	Α	0° - 30°	А	
Change of course		Changing course less than 45° A Changing course less than 45°			Α	
Cascade occurs		No	Α	No		

12. High angle of attack recovery - 4.4.12									
Recovery	Spontaneous in less than 3 sec				А	Spontaneous in less than 3 sec			
Cascade occurs		No			Α	No			Α
13. Recovery from a developed full stall - 4.4.									
Dive forward angle on exit		0° - 30°			A A	0° - 30° No collapse			A
Collapse Cascade occurs (other than collapse)		No collapse No			A	No collapse			A
Rocking backward Line tension		Less than 45° Most lines tight		-	A A	Less than 45° Most lines tight			A A
14. Asymmetric collapse (trim speed) - 4.4.14	Wost lines tight			А	wost lines tight			А	
Folding lines used		No				No			
Change of course until re-inflation	trim speed, max 50% collapse	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	0° - 15°	Α
Re-inflation behavior		Spontaneous re-	-inflation		Α	Spontaneous re	-inflation		Α
Total change of course	trim speed c 50% colls	Less than 360° No			A	Less than 360°	A		
Collapse on the opposite side occurs	trim ax 50				Α	No	Α		
Twist occurs Cascade occurs	ε	No No			A	No No			A
Change of course until re-inflation		< 90°	Dive or roll angle	15° - 45°	A	< 90°	Dive or roll angle	15° - 45°	A
-	trim speed, max 75% collapse			10 10				10 10	
Re-inflation behavior	speed,	Spontaneous re-	-inflation		Α	Spontaneous re	-inflation		Α
Total change of course	trim s x 75%	Less than 360°			A	Less than 360° No No			A
Collapse on the opposite side occurs Twist occurs	max	No			Α				Α
Cascade occurs					Α	No	Α		
Change of course until re-inflation	m	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	0° - 15°	Α
	accelerated, max 50% collapse		inflatio-	1			inflatic -	1	
Re-inflation behavior	accelerated, x 50% colla	Spontaneous re-	-iritiation		A	Spontaneous re Less than 360°	-iritiation		A
Total change of course Collapse on the opposite side occurs	acce x 50°	Less than 360° No			A	No No			A
Twist occurs	na.	No			Α	No			Α
Change of course until as inflation		No		4F0 (F1	A	No		150 150	A
Change of course until re-inflation	esd	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	accelerated, max 75% collapse	Spontaneous re-	-inflation		Α	Spontaneous re	-inflation		Α
Total change of course	celer 75%	Less than 360° No			Α	Less than 360°			Α
Collapse on the opposite side occurs Twist occurs	ac nax j				A	No No	A		
Cascade occurs					A	No			A
15. Directional control with a maintained asymmetry	metric col	-							
Able to keep course straight					A	Yes	A		
180° turn away from the collapsed side possible in	10 sec	Yes			Α	Yes			Α
Amount of control range between turn and stall or spin More than 50% of the			of the symmetric of	control travel	Α	More than 50%	Α		
16. Trim speed spin tendency - 4.4.16		<u> </u>				<u> </u>			1
Spin occurs		No			Α	No	Α		
17. Low speed spin tendency - 4.4.17					L vi.				
Spin occurs 18. Recovery from a developed spin - 4.4.18		No			Α	No	Α		
		Stone opinion	n loce than 000		^	Stops opinion	^		
Spin rotation angle after release		Stops spinning in less than 90°			A	Stops spinning i	A		
Cascade occurs 19. B-line-stall - 4.4.19		No A			А	No	Α		
Change of course before release		Changing course	e less than 45°		Α	Changing course	e less than 45°		Α
Behaviour before release		Remains stable with straight span			Α	Remains stable with straight span			Α
					A				
,	Recovery		Spontaneous in less than 3 sec			Spontaneous in	A		
Dive forward angle on exit Cascade occurs		0° - 30° No			A	0° - 30° No		A A	
20. Big ears - 4.4.20									
Entry procedure	Special device required			Α	Special device r	equired		Α	
Behaviour during big ears		Stable flight			Α	Stable flight			Α
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in less than 3 sec			Α
Dive forward angle on exit	0° - 30°			А	0° bis 30°			A	
21. Big Ears in accelerated flight - 4.4.21									
Entry procedure	Special device required			Α	Special device required			Α	
Behaviour during big ears		Stable flight		Α	Stable flight			Α	
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in	Α		
Dive forward angle on exit Behaviour immediately after releasing the accelarator while		0° - 30°		Α	0° bis 30°			Α	
		Stable flight			Α	Stable flight			Α
maintaining big ears 23. Alternative means of directional control -4.4.22									
					^	Vos			^
180° turn achievable in 20 sec				A	Yes			A	
Stall or spin occurs 23. Any other flight procedure and/or configura	ation deed	No cribed in the user	's manual - 4 4 9	23	А	No			Α
Procedure works as descibed			4.4.		NA				NA
Procedure suitable for novice pilots Cascade occurs			NA				NA		
24. Remarks of testpilot:					NA				NA
		L				L			

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