FTR - Flight Test Report

Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nicht auszugsweise, vervielfältigt werden.





Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing 07.05.	.2015	Minimum take of 62 kg	weight	Maximum take off weight 85 kg			
Testpilot		Mike Küng		Hannes Tschofen			
Harness		EAPR-Testequipment		EAPR-Testequipment			
Pilot's take off weight		65 k	g	85 kg			





Test-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation			
1. Inflation / take-off - 4.4.1		•						
Rising behavior		Easy rising, some pilot correction is required	В	Easy rising, some pilot correction is required	В			
Special take off technique required		No	Α	No	Α			
2. Landing - 4.4.2								
Special landing technique required		No	Α	No	Α			
3. Speeds in straight flight - 4.4.3								
Trim speed more than 30km/h		Yes	Α	Yes	Α			
Speed range using the controls larger than 10km/h		Yes	A	Yes	А			
Minimum speed		25 km/h to 30 km/h	В	25 km/h to 30 km/h	В			
4. Control movement - 4.4.4		20 14111 10 00 14111		20 11111110 00 111111				
			1	Γ				
Max. weight in flight up to 80kg	ax. weight in flight up to 80kg		-		-			
Max. weight in flight 80 to 100kg	ax. weight in flight 80 to 100kg		D	Increasing 35cm - 45cm	D			
Max. weight in flight greater than 100kg	Max. weight in flight greater than 100kg				-			
5. Pitch stability exiting accelerated flight - 4.	4.5							
Dive forward angle on exit		Dive forward less than 30°	Α	Dive forward less than 30°	Α			
Collapse occurs No			А	No A				
6. Pitch stability operating controls during acc	elerated 1	•		1				
Collapse occurs		No	А	No A				
7. Roll stability and damping - 4.4.7								
Oscillations		Reducing	Α	Reducing A				
8. Stability in gentle spirals - 4.4.8								
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α				
9. Behaviour exiting a fully developed spiral d	ive - 4.4.							
Initial response of glider (first 180°)		No immediate reaction	В	No immediate reaction	В			
Tendency to return to straight flight Turn angle to recover normal flight		Spontaneous exit 720° to 1080°, spontaneous recovery	A B	Spontaneous exit Less than 720°, spontaneous recovery	A			
-		720 to 1080 , spontaneous recovery	Б	Less than 720 , spontaneous recovery	А			
10. Symmetric front collapse - 4.4.10		T.		Lw				
Folding lines used Entry		No Rocking back less than 45°	Α	No Rocking back less than 45°	Α			
	~ 30%							
Recovery		Spontaneous in 3 to 5 sec	В	Spontaneous in less than 3 sec	Α			
Dive forward angle on exit	rim speed	30° - 60° Entering a turn of less than 90°	В	30° - 60° Keeping course	В			
Cascade occurs		No	A	No No	A			
Entry	> 50%	Rocking back less than 45° Recovery through pilot action in less than a furthe	A	Rocking back less than 45°	A			
Recovery		3 sec	D	Spontaneous in 3 to 5 sec	В			
Dive forward angle on exit	rrim speed	30° - 60° Entering a turn of 90° to 180°	С	30° - 60° Entering a turn of 90° to 180°	С			
Cascade occurs		No	Α	No	Α			
Entry	20%	Rocking back less than 45°	Α	Rocking back less than 45°	Α			
Recovery	accelerated > 4	Recovery through pilot action in less than a furthe 3 sec	D	Spontaneous in 3 to 5 sec	В			
Dive forward angle on exit	ejeoo	30° - 60° Entering a turn of 90° to 180°	- J		C			
Cascade occurs		No	А	No	Α			
11. Exiting deep stall (parachutal stall) - 4.4.1	1	T		1				
Deep stall achieved		Yes		Yes				
Recovery		Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec				
Dive forward angle on exit		30° - 60° Changing course less than 45°	В	30° - 60°	В			
Change of course			A	Changing course less than 45°	A			
Cascade occurs		No	Α	No	Α			

12. High angle of attack recovery - 4.4.12									
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in less than 3 sec			А
Cascade occurs		No		Α	No No			Α	
13. Recovery from a developed full stall - 4.4.1									
Dive forward angle on exit		60° - 90° No collapse			C A	60° - 90°			C A
Collapse Cascade occurs (other than collapse)		No			A	No collapse			A
Rocking backward	Rocking backward				A	Greater than 45)		C
Line tension 14. Asymmetric collapse (trim speed) - 4.4.14		Most lines tight			Α	Most lines tight			Α
Folding lines used	No				No				
Change of course until re-inflation	ө	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-	inflation		Α	Spontaneous re-	inflation		Α
Total change of course	speed, % colla	Less than 360°			A	Less than 360°	A		
Collapse on the opposite side occurs	trim ax 50°	No No			A	No No			A
Twist occurs Cascade occurs	Ë	No No			A	No No			A
Change of course until re-inflation		90° - 180°	Dive or roll angle	15° - 45°	В	< 90°	Dive or roll angle	45° - 60°	C
	trim speed, max 75% collapse							43 - 00	_
Re-inflation behavior	trim speed < 75% colla		nan 3 sec from sta	art of pilot action	С	Spontaneous re-	inflation		A
Total change of course Collapse on the opposite side occurs	trim 8 x 759	Less than 360°			A	Less than 360° No No			A
Twist occurs	max	No		Α	Α				
Cascade occurs		No		1	Α	No			Α
Change of course until re-inflation	se	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-	inflation		А	Spontaneous re-	inflation		А
Total change of course	accelerated, ıx 50% collap	Less than 360°			Α	Less than 360°			А
Collapse on the opposite side occurs	acı	No			A	No			A
Twist occurs Cascade occurs	_	No No			A	No No			A A
Change of course until re-inflation	m	90° - 180°	Dive or roll angle	15° - 45°	В	< 90°	Dive or roll angle	45° - 60°	С
Re-inflation behavior	accelerated, max 75% collapse	Inflates in less th	nan 3 sec from sta	art of pilot action	С	Spontaneous re-	inflation		A
Total change of course	accelerated, x 75% collap	Less than 360°	ian 5 sec nom se	art or prior action	A	Less than 360°	illiation		A
Collapse on the opposite side occurs	acce ax 75	No			A	No			A
Twist occurs	Ĕ	No No			A	No No			A A
Cascade occurs 15. Directional control with a maintained asymmetry	netric col				A	140			A
Able to keep course straight	Yes			Α	Yes			А	
180° turn away from the collapsed side possible in	10 sec	Yes			Α	Yes			Α
Amount of control range between turn and stall or spin		25% to 50% of the symmetric control travel		С	25% to 50% of the symmetric control travel			С	
16. Trim speed spin tendency - 4.4.16		25% to 56% of the symmetric control naver							
Spin occurs	No			Α	No	Α			
17. Low speed spin tendency - 4.4.17									
Spin occurs		No	40		Α	No			Α
18. Recovery from a developed spin - 4.4.18								_	
Spin rotation angle after release		Stops spinning in less than 90°		Α	Stops spinning in 90° to 180°			С	
Cascade occurs 19. B-line-stall - 4.4.19	No			Α	No	Α			
Change of course before release		Changing course	e less than 45°		Α	Changing course	e less than 45°		Α
Behaviour before release	v		Unstable		D	Remains stable without straight span			С
_									
Recovery		Spontaneous in less than 3 sec		A	Spontaneous in 3 to 5 sec			В	
Dive forward angle on exit Cascade occurs	Dive forward angle on exit		30° - 60° No		A	0° - 30° No			A
20. Big ears - 4.4.20									
Entry procedure		Standard technique		А	Special device required			А	
Behaviour during big ears		Stable flight		Α	Stable flight			А	
Recovery		Spontaneous in 3 to 5 sec		В	Recovery through pilot action in less than a further			В	
Dive forward angle on exit	0° - 30°			Α	3 sec 0° bis 30°			А	
21. Big Ears in accelerated flight - 4.4.21									
Entry procedure		Standard technique		Α	Special device required			Α	
Behaviour during big ears		Stable flight		Α	Stable flight			Α	
Recovery		Spontaneous in 3 to 5 sec		Α	Recovery through pilot action in less than a further 3 sec			В	
Dive forward angle on exit		0° - 30°		Α	0° bis 30°			Α	
Behaviour immediately after releasing the accelarator while maintaining big ears		Stable flight		Α	Stable flight			Α	
23. Alternative means of directional control - 4.4.22									
180° turn achievable in 20 sec	Yes			Α	Yes			Α	
Stall or spin occurs	No			Α	No				
23. Any other flight procedure and/or configuration described in the user's manual - 4.4.23									
Procedure works as descibed Procedure suitable for novice pilots				NA NA					
Cascade occurs	NA NA				NA NA				
24. Remarks of testpilot:					1				
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